

# Lunar Lander / Orbiter CubeSats

Dr. Carl Brandon

carl.brandon@vtc.edu

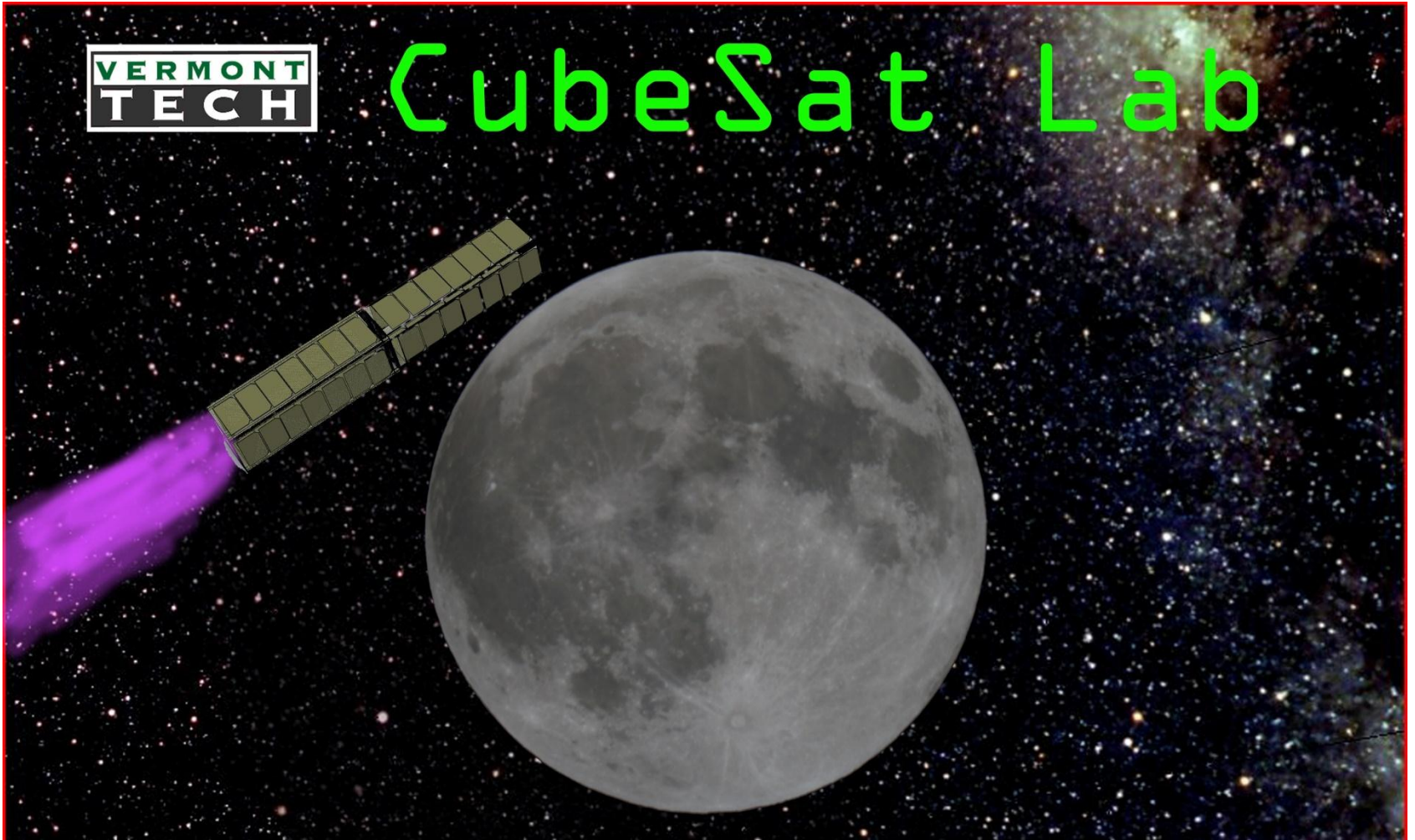
Vermont Technical College

+1-802-356-2822 (Voice)

Randolph Center, VT 05061 USA <http://www.cubesatlab.org>



# CubeSat Lab



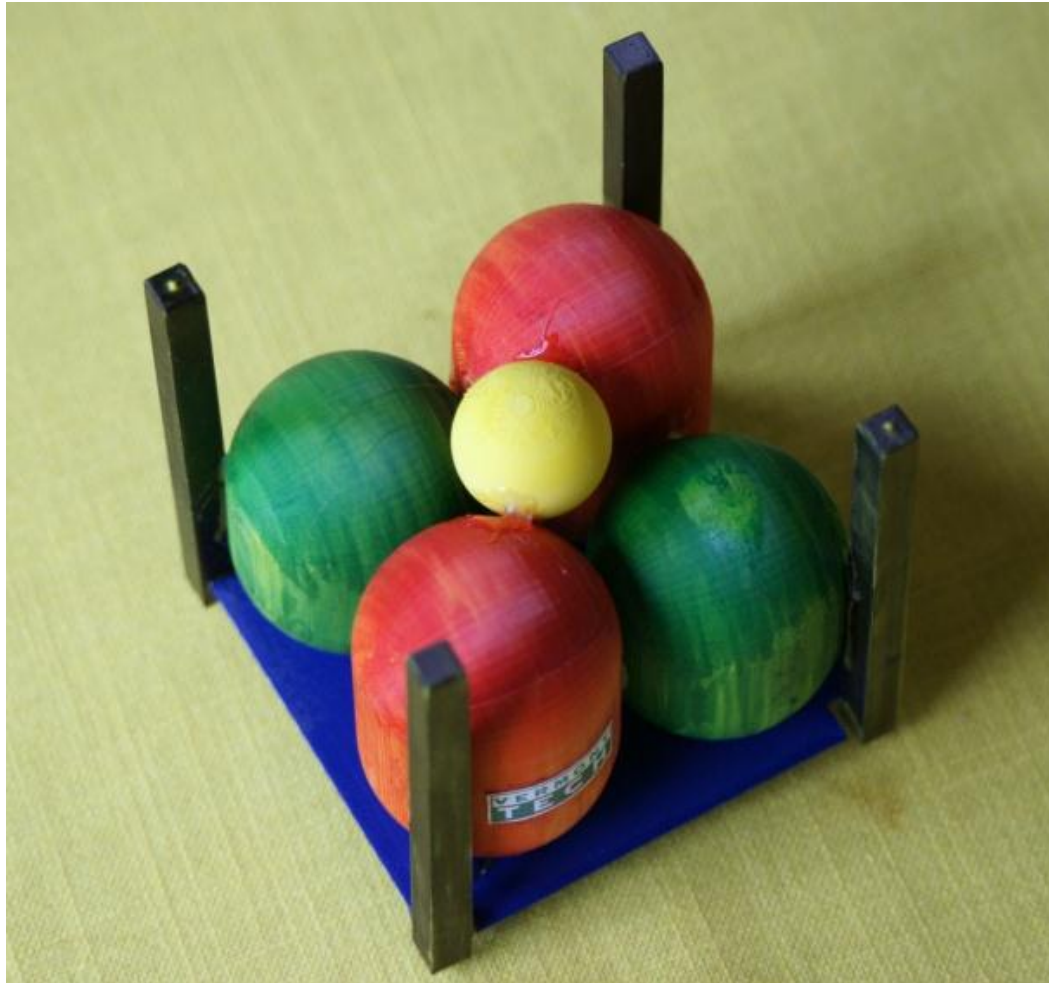
# NASA Consortium Development Grant

- Vermont Technical College: Carl Brandon (structure, thrusters, communication, navigation, electronics) and Peter Chapin (software)
- University of Vermont: Jun Yu (low energy transfer and radiation environment modeling) and Jeff Frolik (spacecraft coordination, landing)

# NASA Consortium Development Grant

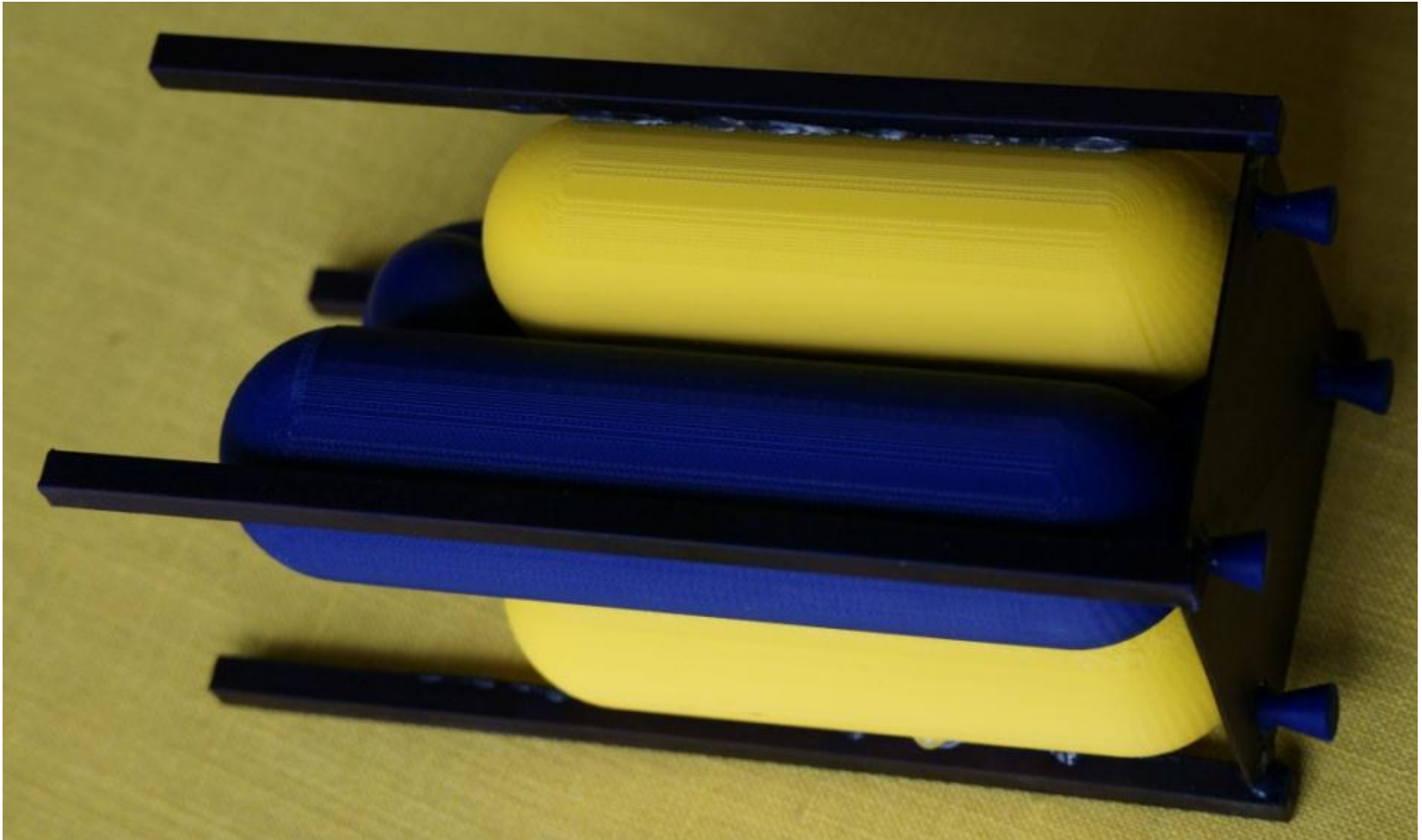
- Norwich University: Ron Lessard and Danner Friend (optical sensors and robotics)
- Undergraduate students and graduate students (UVM) from the above institutions and St. Michael's College
- Assistance from NASA Goddard Spaceflight Center and NASA Jet Propulsion Laboratory personnel

# Three Spacecraft



Single CubeSat Lunar Lander

# Three Spacecraft



Double CubeSat Booster

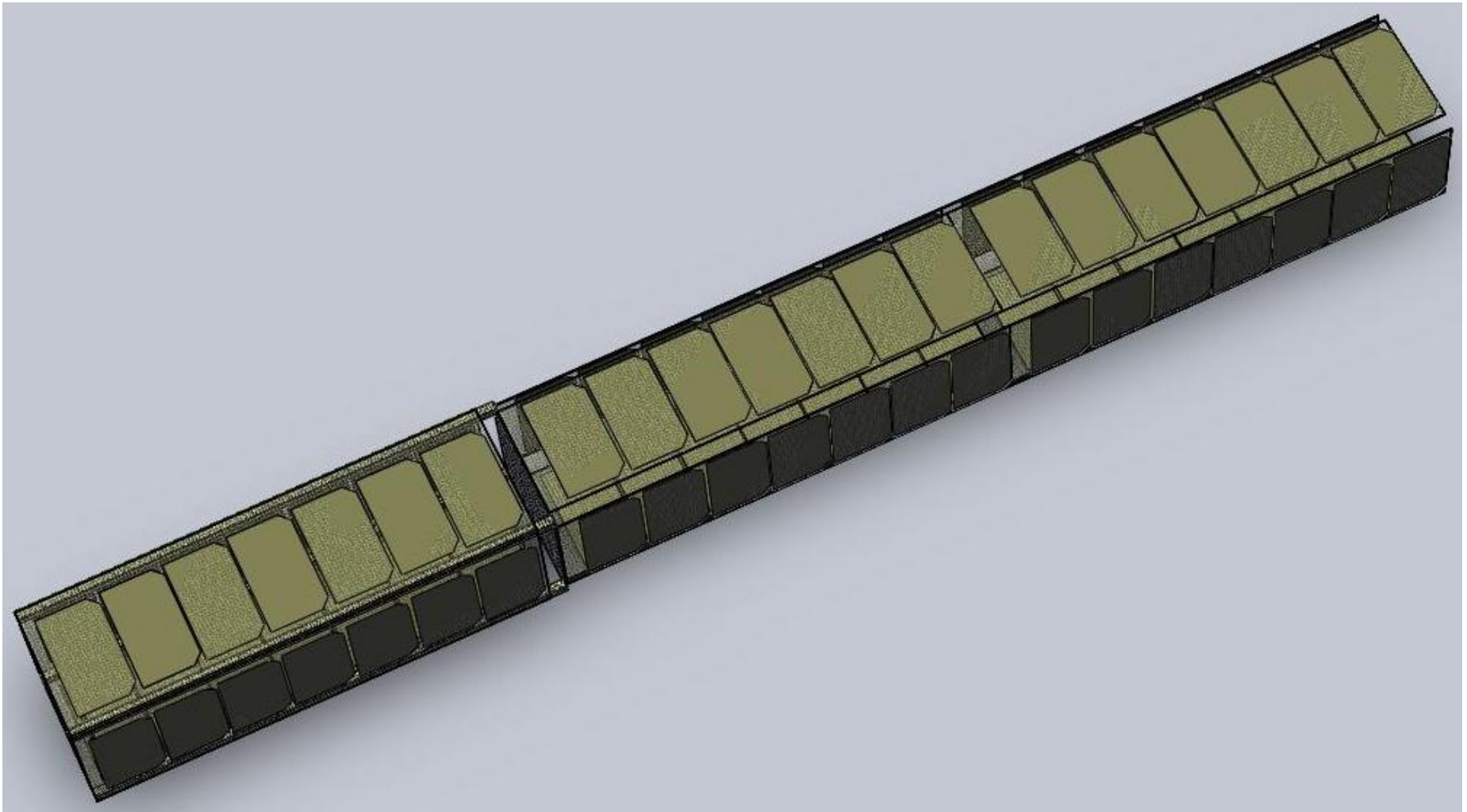


# Three Spacecraft



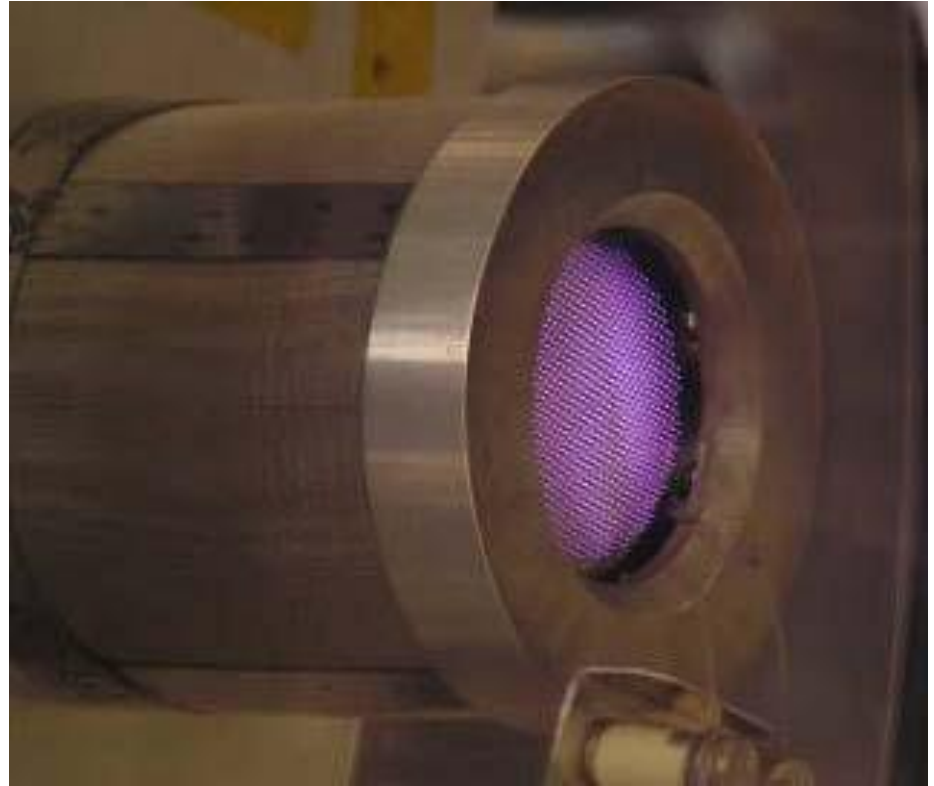
Triple CubeSat Ion Lunar Orbiter

# Three Spacecraft



Higher Power Triple CubeSat Ion Lunar Orbiter

# Three Spacecraft



JPL Miniature Xenon Ion (MiXI) Thruster



# Technology Prototyping

- Use of carbon fiber composite for CubeSat structures
- Development of a 1N mono-propellant thruster
- Designing a CubeSat sized thruster gimbal
- Developing a CubeSat sized 200-300 atmosphere xenon tank

# Technology Prototyping

- Miniaturizing the xenon pressure regulator and gas flow control as used in the ESA SMART-1 spacecraft
- Converting the NASA Goddard GEONS navigation system to Ada/SPARK and developing the associated hardware
- Developing the associated electronics and sensors
- Developing optical means for lunar landing attitude control and lateral velocity control

# Chemical Propulsion System

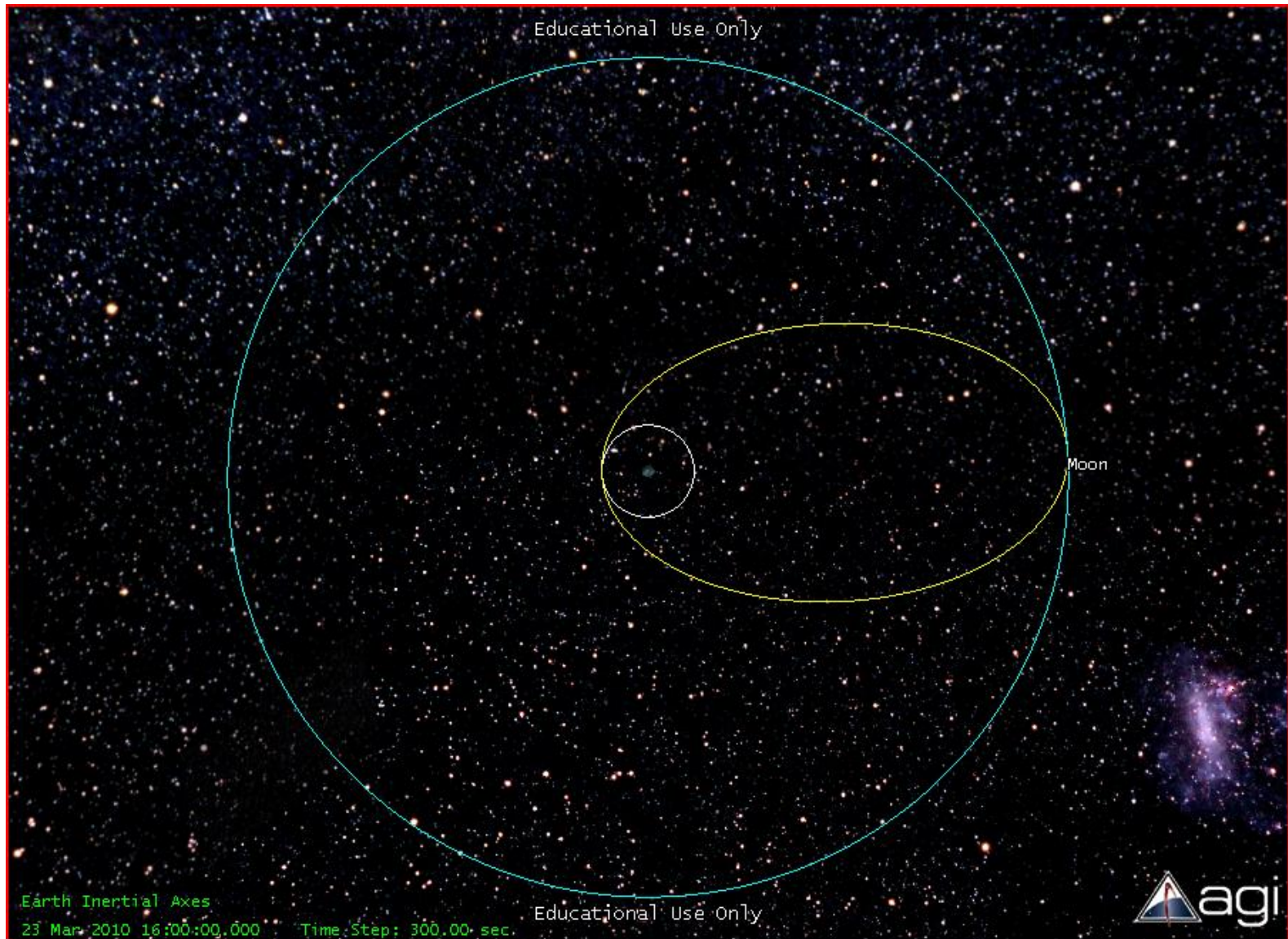
- Lander and Booster will each have four 1 N thrusters
- Monopropellant: Hydroxyl Ammonium Nitrate and Methanol (88%:12% by mass) mixed
- Specific Impulse about 270 seconds

# Ion Propulsion System

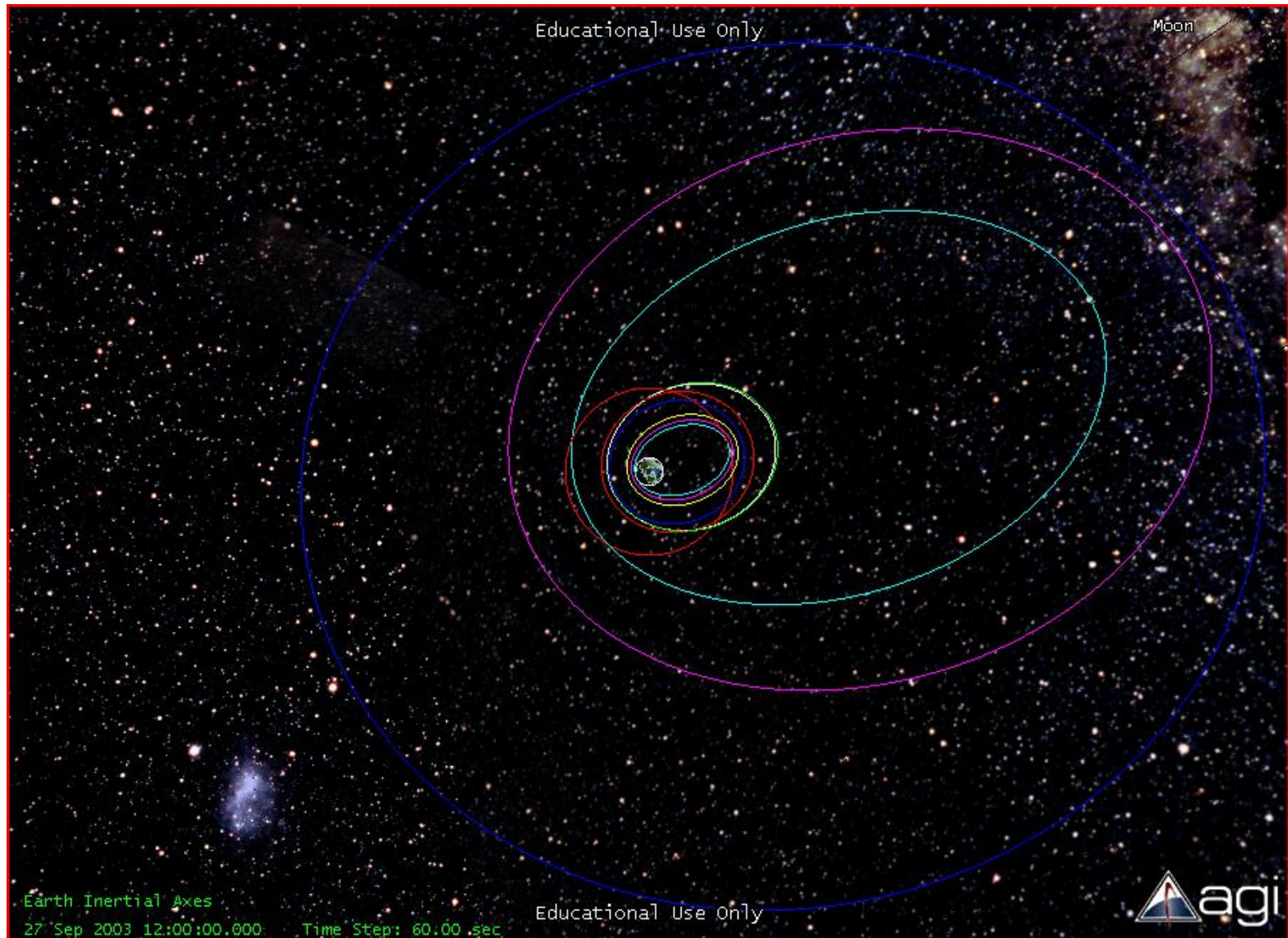
- Triple CubeSat orbiter will have one MiXI 1 mN thruster, Specific Impulse about 2,500 – 3,000 seconds
- Power for the xenon ion drive will come from photovoltaic cells on the spacecraft body and four fold out 30 cm x 10 cm panels



# Bi-Propellant Orbits



# Ion Drive Orbits ala SMART-1





# Vermont Tech Student Involvement

- Electro-Mechanical Engineering Technology Bachelors students can take the Spacecraft Systems course and work in the CubeSat Lab, on the Lunar spacecraft which becomes their required Senior Project
- They implement the specific systems for the CubeSat.

# Vermont Tech Student Involvement

- Software Engineering Bachelors students would implement parts of the software systems for their Senior Project and are currently developing a test framework for the navigation system.
- Other Electro-Mechanical students have worked on a steerable dish antenna for the GENSO ground station and a testing system for the Lunar lander landing gear



# Vermont CubeSat Project

- Carbon fiber composite for structural components (including fold out photovoltaic panels)
- Using a CubeSat Kit CPU board
- Texas Instrument MSP430 CPU

# Vermont CubeSat Project

- Spectrolab TASC 27% efficient triple junction photovoltaic cells
- Clyde Space Electrical Power System with Li-Polymer batteries
- 3 axis magnetometer (near earth) and sun, earth, moon sensor for attitude determination
- Differential thrusters (chemical), and gimbaled ion thruster for attitude control

# Vermont CubeSat Project

- NASA Goddard GPS Enhanced Onboard Navigation System (GEONS) for navigation (rewritten in Ada/SPARK)
- Radios for 2-way communication to be determined
- Ground communication station to be part of the worldwide GENSO network
- Camera modules to be used for celestial navigation.

# Why Use Ada and SPARK?

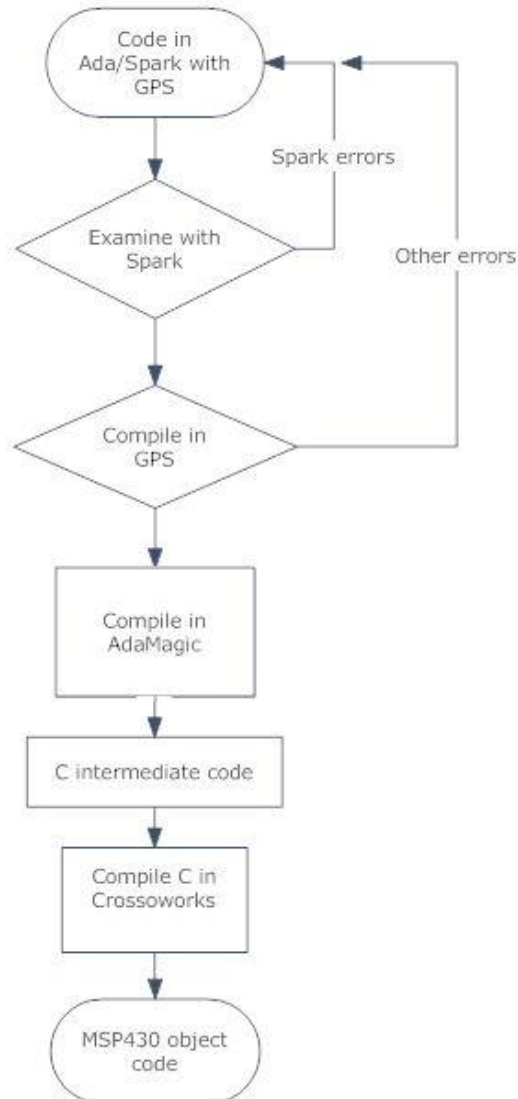
- Although not safety critical, the software is mission critical. We won't be able to patch it.
- Satellite cost of \$100,000-\$300,000 and launch cost of \$200,000-\$300,000 and several years' effort would be lost by a software failure
- Ada/SPARK offers a greatly improved probability of error-free software when compared with C used in most CubeSat projects



# Why Use Ada and SPARK?

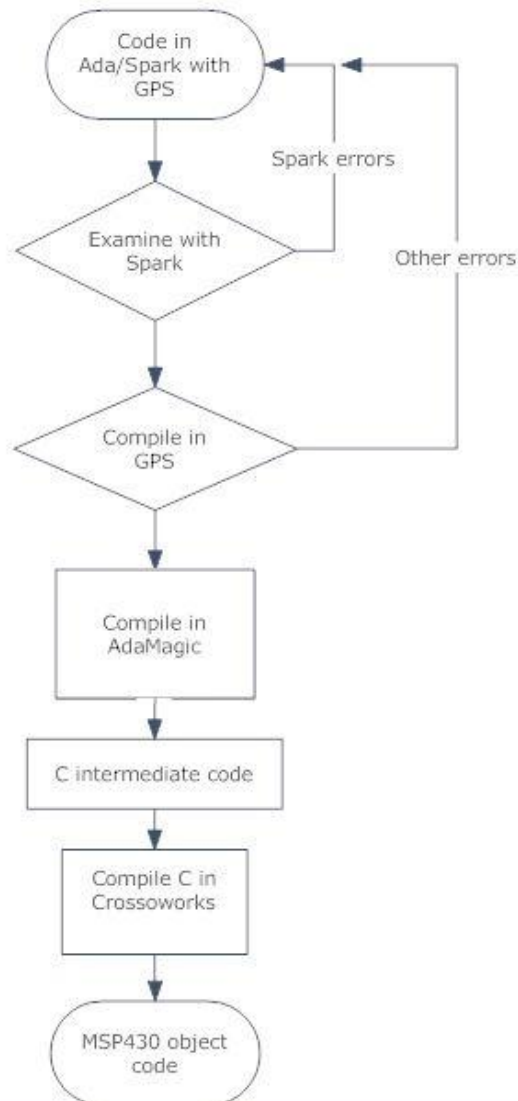
- The development and debugging time would be less; helpful with our smaller resource base
- Students get a chance to use the best software engineering features in Ada and SPARK in a real-world embedded system

## Software Development Process



- Software written in Ada/SPARK
- Checked with the Praxis High Integrity Systems' SPARK Toolset
- Compiled with Adacore's GNAT Pro

## Software Development Process



- Sofcheck's AdaMagic compiles to produce C intermediate code
- Rowley's Crossworks C to object code for Texas Instruments' MSP430 CPU
- Pumpkin's Salvo RTOS if needed

# Acknowledgements

- NASA Vermont Space Grant Consortium



- NASA



- Vermont Technical College



- AdaCore, Inc.



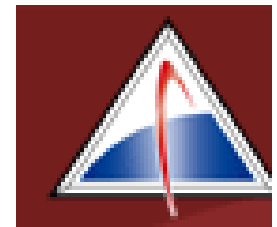
- Praxis High Integrity Systems



- SofCheck



- Applied Graphics, Inc.



# Vermont's Low Cost Astronaut





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